

Investigation of Flexural Properties of Glass-Kevlar Hybrid Composite

Y. M. Kanitkar, A.P. Kulkarni, and K. S. Wangikar

Abstract—Composite materials are attracting huge attention due to their superior properties and being inert to most atmospheric effects. They have high strength to weight ratio and can be moulded into the required shape that can be used for various applications as replacement for metals. This paper consists of study of flexural behaviour of hybrid composite reinforced with woven glass fiber and Kevlar fiber in ply configuration. The three point bending test according to ASTM 790 was performed experimentally on this composite laminate. The fiber volume fraction to matrix volume fraction was taken 40%-60%. The volume fraction of the individual fibers in the composite was varied to determine the effects on the flexural strength of the composite laminate. The laminates prepared were having dimension 80mm X 13mm X 3mm. After testing the H4 configuration had the highest flexural strength of 217.91 MPa also the H2 configuration had maximum flexural modulus. The result of the study can be used for the development of actual composites and simulation purpose.

Index Terms—Hybrid Composite, Kevlar Fiber, Glass Fiber, Flexural Strength.

I. INTRODUCTION

Composite materials are replacing the metals as an alternate material for various applications as they have high strength to weight ratio. The composites are made of fiber reinforcement bound by a matrix material [1]. The composites can be made from natural fiber and manmade fibers. The use of the natural fiber composites is limited to secondary applications and they possess less strength and they being biodegradable are not suitable for primary applications. This has led to the use of manmade materials which have high strength and are not susceptible to atmospheric effects [2]-[3].

Most famous manmade fibers used for making composites are Glass, Carbon, Kevlar, Asbestos, Ceramics etc. out of this Glass Carbon and Kevlar fibers are most prominently used for automobile body parts, sports goods, marine application, bullet-proofing, civil structures electric insulations, heat resistant clothes etc. [4]. The composites

made out of any one fiber is called pure composites which gives specific characteristic like composite made from Carbon or Kevlar fibers have superior strengths but are very costly whereas the composite made from Glass fibers are cheap but their strengths is less. So to achieve the balance between the strengths and cost of composite, two or more fibers are being mixed to make composite this type of composites are called hybrid composites. The hybrid composites have the advantage of low cost and strengths greater than the weaker material but less than the stronger material. Essentially the hybrid composite can be defined as a composite consisting two or more different kinds of material as its reinforcement [5]-[7].

Many experimental works are being carried out on the hybrid composite for more than 30 years to study the effects of hybridization on the properties of the composite [8]-[16]. Aramide fibers have drawn lot of focus as it is heat resistant and high strength fibers, the fibers have tensile strength of 2.8 GPa and modulus of elasticity of 109 GPa. Functional aramides have found their way into various applications as industrial materials, bulletproof and protective shields, marine fishery parts and civil structures [17]-[20].

The fiber volume fraction plays an important role in determining the strength of the composite prepared. Rejab et al. investigated the effects of fiber volume fraction on GFRP Plate. The volume fractions used for the experimentation were 20-80%, 30-70%, 40-60% Fiber to Matrix volume ratio respectively. The optimum tensile and compressive strength was observed in the case for 40%-60% fiber to matrix volume ratios. It was seen that as the fiber volume increased in the composite the strengths also increased but after certain limit as fiber volume fraction increases the amount of matrix material decreases and it starts weakening interfacial bonding strength between fibers and binder material [21].

Dong et al. investigated the effect of different volume fractions of S-2 glass and T700S carbon fiber on the flexure strength of the composite. The optimum hybrid effect was achieved when the hybrid ratio was 0.125 ([0G/07C]) when both V_{fc} and V_{fg} were 50% and the increases in strength was 43.46% and 85.57% when compared with those of the full carbon and glass configurations respectively. The optimisation shows that the maximum hybrid effect was 56.1% when V_{fc} = 47.48% and V_{fg} = 63.29% [22].

Along with the fiber volume fraction the fiber orientation also play an important role for change in mechanical properties. Various experiments were carried out using different fiber orientation like 0°, 90°, +45°, -45° fiber orientation in the composite laminate and the 90° and +45°

Manuscript published July 31, 2016.

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orientation gave maximum tensile strengths out of all the combinations [23]-[25].

Many researchers have tested the mechanical properties of composite having different layering sequences and results obtained proved that layering sequence affected the mechanical properties of the composites. For instance Sarasini et al. prepared the composite having sandwich-like stacking sequence and the other one with an intercalated fabrics lay-up for carbon laminates and performed low velocity impact testing and observed hybrid laminates with

Sandwich configuration failed by interface failures located in the basalt skins of the compression side which prevented hindering crack propagation in the carbon core. Another researcher Zhang et al. used Glass-Carbon hybrid composite having different stacking sequence to determine the mechanical properties like tensile, compression and three point bending and it was found that the hybrid containing 50% carbon fibers showed improvement in the flexure properties when carbon fibers are kept at exterior region, and alternating carbon/glass layup gave good compressive strength and tensile strength was not affected by the stacking sequence. [26]-[29].

Flexural strength is usually measured using three point bending test. For this test when loading is done the upper side of the specimen will experience compression and the lower side will experience tension and at mid plane the specimen will experience shear stress. Using this principle when Prusty et al. [30] replaced four glass plies out of seven ply composite laminate by carbon it was found that by the carbon at tensile side showed improvement in strengths and modulus but the laminate was more prone to catastrophic failure whereas at the compressive side the carbon epoxy showed progressive failure also the laminate had 93% modulus and 96% strength as that of laminate made entirely of carbon epoxy.

Another researcher Dongs [31] choose two types of glass S-2 and E, and two Carbon fiber T700S and P-100 and determined the flexural strength using three different models namely Lo-Chim, Budiansky and the shear model. The flexural strength obtained by first two models was nearly similar but the flexural strength obtained by shear model was 20-30% lower. And Sudarisman and Davies [32] noted that when E-glass fibers were replaced by 33% of S-2 glass fibers the flexural strength increased by 23%, and it did not have any significant effect on flexural modulus. Kalnin [33] found that the flexural strength decreases rapidly as all-glass reinforcement is progressively replaced by graphite fibre. Slight positive deviations were shown from the theoretical maximum strain failure criteria.

II. EXPERIMENTAL DETAILS

A. Materials

The laminates prepare for mechanical testing consist of Kevlar fiber 200 GSM and Glass fiber 200GSM having 0° and 90° orientation. The Kevlar fibers have density of 1.40 g/cc which was taken from the manufacturer's catalogue as shown in Table I. The glass fiber in form of woven mat was used and the density of Glass fiber mat is 1.90 g/cc shown in Table II. The liquid type XR-125 epoxy resin having density 1 is used for preparing the laminates. The details of which

are given in Table III. Fig. 1, Fig. 2 and Fig. 3 shows the Glass fiber, Kevlar Fiber and Epoxy Resin and Hardener used for manufacturing the composite

TABLE I: PROPERTY OF KEVLAR FIBERS

Property	Unit	Value
Width	cm	150
Weight	g/m ²	192
EPI (Warp)	ea	21
EPI (Weft)	ea	20.8
Breaking Strength (Warp)	N/5cm	8647
Breaking Strength (Weft)	N/5cm	7887
Thickness	mm	0.3

TABLE II: PROPERTY OF GLASS FIBERS

Property	Units	Value
Single Filament Tensile Strength	MPa	3100-3800
Single Filament Tensile Strength	Kpsi	450-550
Young's Modulus of Elasticity	GPa	80-81
Young's Modulus of Elasticity	Mpsi	11.6-11.8
Fiber Density	g/cc	2.62

TABLE III: PROPERTY OF EPOXY RESIN

Property	Unit	Value
Tensile Strength	MPa	50-60
Flexural Strength	MPa	120-130
Elastic Modulus in Flexural	GPa	3-3.5



Fig. 1. Glass Fiber Cloth



Fig. 2. Kevlar Fiber Cloth



Fig. 3. Epoxy Resin Used



Fig. 4. Preparation of Laminate



Fig. 5. Application of Pressure using Compression Molding Machine

B. Composite Laminate Preparation and sample configuration

The Glass fiber and the Kevlar fiber laminates having epoxy resin as the matrix material were prepared using the hand-lay-up method. The Glass and Kevlar fiber mats were hand laid-up using epoxy matrix as a binder which contained the mixture of epoxy resin and hardener in the ratio of 10:1 as shown in Fig. 4. Total six laminates were prepared with the configuration as given in the Table IV. The laminates were made using the compression molding machine and covering plastic was used on the outer surface of the laminate so the epoxy matrix would not stick to the metal plates of the compression molding machine and it also provided smooth and even outer surface it as shown in Fig. 5 and Fig. 6 shows the manufactured composite laminate.

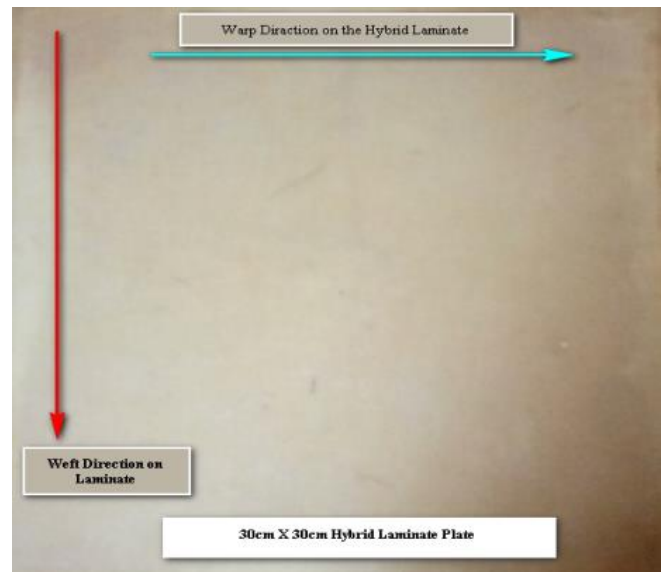


Fig. 6. Manufactured laminate of size 30cmX30cmX0.3 cm

TABLE IV: FIBER VOLUME FRACTION AND FIBER SEQUENCE

Laminate Number	Volume Fraction of Fiber to Matrix	Number of Glass Fiber Layers	Number of Kevlar Fiber Layers	Volume Fraction of Glass to Kevlar fiber in 40% fiber volume	Sequence
H1	40%-60%	5	4	21%-19%	G-K-G-K-G-K-G-K-G
H2	40%-60%	6	3	25%-15%	G-K-G-G-K-G-G-G-K-G
H3	40%-60%	7	2	30%-10%	G-G-G-K-G-K-G-G-G
H4	40%-60%	4	5	16%-24%	G-K-G-K-K-K-K-G-K-G
H5	40%-60%	3	6	12%-28%	G-K-K-K-G-K-K-K-G
H6	40%-60%	2	7	7%-33%	G-K-K-K-K-K-K-K-G

C. Fiber Volume Fraction Calculation

It is defined as the amount of fiber contained in the composite. Higher the volume fraction of fibers in the composite results in the enhanced mechanical properties but there is certain limitation that is after certain limit as fiber volume fraction increases the amount of matrix material

decreases and it starts weakening interfacial bonding strength between fibers and binder material. So to achieve best mechanical properties the fiber volume fraction is kept 40% in the laminate and the volume fraction is calculated using the following formula: [22]

$$V_f = \frac{W_f/\rho_f}{W_m/\rho_m + W_{f1}/\rho_{f1} + W_{f2}/\rho_{f2}} * 100 \quad (1)$$

Where

V_f = Fiber Volume Fraction

W_f = Weight fraction of Fiber

ρ_f = Density of Fiber

W_m = Weight of Matrix

ρ_m = Density of Matrix

W_{f1} = Weight fraction of 1st Fiber

ρ_{f1} = Density of 1st Fiber

W_{f2} = Weight fraction of 2nd Fiber

ρ_{f2} = Density of 2nd Fiber

Also from the literature it is clear that the Glass fibers is better resistant to the atmospheric effects than the Kevlar fiber and it being the material having very high strength is the most suitable material for making the core portion the laminate. The arrangement of the Glass fibers and Kevlar fibers along with their respective volume fraction inside the composite laminate and the total volume fraction of the fiber to matrix is given in the Table IV.

D. Flexural Test

Flexural test or three point bending test was performed to determine the stress-strain relationship and the flexural modulus of the composite laminate having different combination of Glass and Kevlar fibers was carried out according to ASTM D790. The specimen were cut using water jet machining and the dimensions of length by width by thickness were 80mm X 13mm X 3mm respectively. The specimen was tested on Universal Testing Machine with model no- STS 248, having speed 5mm/min and with the accuracy of $\pm 1\%$. The flexural test specimen and testing of specimen is shown in the Fig 5.

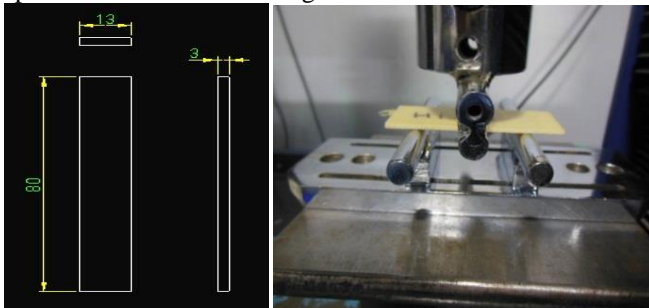


Fig. 5. Flexural Test Specimen Dimensions and Flexural Testing of Specimen

III. RESULTS AND DISCUSSIONS

The results obtained after performing the experiments on

the specimen are summarized in the Table V. Total five tests were carried out for one configuration and the average of all the five test result are summarized in the table.

Specimen Configuration	Flexural Strength(MPa)	Flexural Modulus(Gpa)
H1	196.43	11.016
H2	174.16	15.810
H3	90.23	4.483
H4	217.91	12.821
H5	199.99	13.883
H6	199.99	11.653

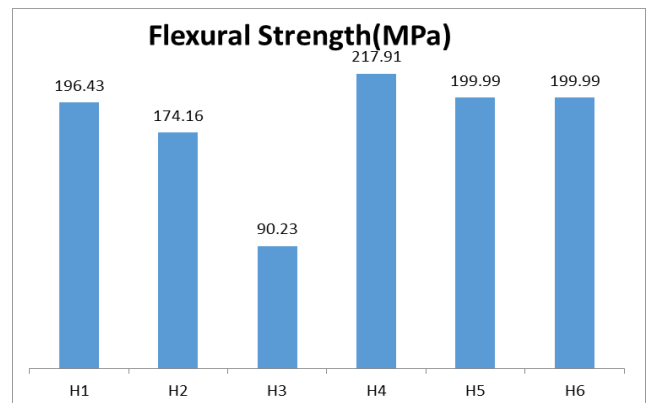


Fig. 6. Application of Pressure using Compression Molding Machine

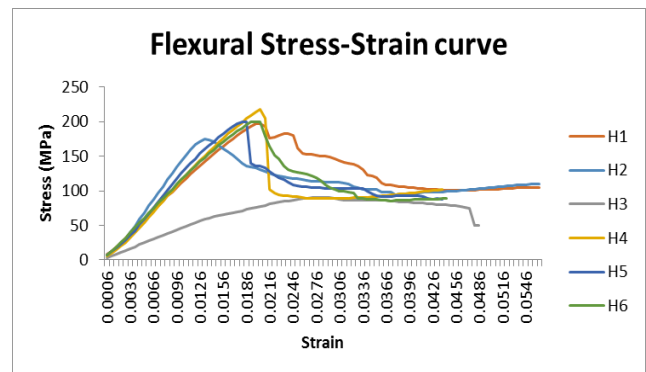


Fig. 7. Flexural Stress-Strain Curve

The Table V, Fig. 6 and Fig. 7 shows the values of flexural strengths and also the stress-strain relationship is shown in which gives the flexural modulus. The configuration H4 has the maximum flexural strength of 217.91 MPa, closely followed by H5, H6 and H1 configuration. It can be seen that the introduction of glass fiber with volume fraction 21% and 16% has positive hybridisation effect on the hybrid composite but as the volume of glass fiber increases from 21% to 30% the flexural strength decreases significantly indicating if the high volume of glass fiber leads to negative hybridisation effect on the composite. This same phenomenon happen in case if the Kevlar fiber volume is kept more than the Glass

fiber volume but the drop in strength is not so significant as compare when Glass fiber is dominating the composite. This indicates that the volume fraction of Glass fiber to Kevlar fiber must be kept at 16%-24% respectively.

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